A new single-step reaction mechanism for propane explosions covering the entire spectrum of flame acceleration, transition to detonation and detonation

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Outline

- The new single-step reaction mechanism
- Governing equations & CFD code
- Validation
 - Comparison with detailed chemistry model
 - Detonation cell size
 - Flame acceleration in a vented duct
- Exploratory study shock and detonation propagation through a U-bend
- Concluding remarks



Limitations of existing reaction mechanisms

(1) Berkeley GRI-mechanism (53 species, 325 reactions) computationally intensive

(2) Westbrook's mechanism (1981)
$$-\frac{d[C_{3}H_{8}]}{dt} = 8.6 \times 10^{11} \exp(-15000/T) [C_{3}H_{8}]^{0.1} [O_{2}]^{1.65}$$

under-predicting half reaction length

(3) Frolov's model (2007)
$$-\frac{d[C_{3}H_{8}]}{dt} = 7 \times 10^{14} \times p^{n} \times \exp(-454600/RT)[C_{3}H_{8}][O_{2}]$$

under-predicting half reaction length for rich gas



Our group's previous approach in detonation modelling

Wen, JX, Heidari, A, Ferraris, S and Tam, VHY (2011) Numerical simulation of propane detonation in medium and large scale geometries. *Journal of Loss Prevention in the Process Industries*, 24(2), pp. 187-193. ISSN (print) 0950-4230.

Reaction progress equation:

$$\frac{\partial \rho \alpha}{\partial t} = -\nabla(\rho \alpha V) + \rho \omega \tag{4}$$

$$\omega = A(1-\alpha)EXP(-\frac{E_a}{RT})$$



The predicted overpressure and velocity vs time

Wen, JX, Heidari, A, Ferraris, S and Tam, VHY (2011) Numerical simulation of propane detonation in medium and large scale geometries. *Journal of Loss Prevention in the Process Industries*, 24(2), pp. 187-193. ISSN (print) 0950-4230.







The new single-step reaction mechanism

A singe-step overall reaction for propane-air combustion

$$C_3H_8 + 5O_2 + 18.8N_2 \rightarrow 3CO_2 + 4H_2O + 18.8N_2$$

The reaction rate in Arrhenius form

$$\omega = k [C_3 H_8]^a [O_2]^b$$

where $k = Aexp\left(-\frac{E_a}{RT}\right)$, $[C_3H_8]$, $[O_2]$, . a and b are the rate constant, propane and oxygen molar concentrations, propane and oxygen rate exponents respectively. A and E_a denote pre-exponential factor and activation energy, respectively.



The new single-step reaction mechanism

$$\omega = k [C_3 H_8]^a [O_2]^b$$

Reaction order = a + b3.0 -2.8 2.6 2.4 2.2 $n = -\frac{\rho}{\tau_i} \left(\frac{\partial \tau_i}{\partial \rho}\right)_{\tau_2} + 1$ 2.0 Reaction Order 1.8 1.75 1.6 1.4 1.2 1.0 -S.P.M Bane, J.L.Ziegler and J.E. Shepherd. "Development of One-0.8 Step Chemistry Model for flame and ignition simulation". GALCIT 0.6 Report GALTCITFM:2010.002, 2010 0.4 0.2 0.0 -0.4 0.6 0.8 1.0 1.2 2.0 1.4 1.6 1.8 2.2 2.4 Equavalence Ratio \$

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The new single-step reaction mechanism

$$-\frac{d[C_3H_8]}{dt} = 3.11 \times 10^{14} \exp\left(\frac{-55910}{RT}\right) [C_3H_8]^{0.1} [O_2]^{1.65}$$



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Governing equations

$$\frac{\partial \overline{\rho}}{\partial t} + \frac{\partial \overline{\rho} \widetilde{u}_{j}}{\partial x_{j}} = 0$$

$$\frac{\partial \rho u_{i}}{\partial t} + \frac{\partial \rho u_{i} u_{j}}{\partial x_{j}} = -\frac{\partial p}{\partial x_{i}} + \frac{\partial}{\partial x_{j}} \left[\mu \left(\frac{\partial u_{i}}{\partial x_{j}} + \frac{\partial u_{j}}{\partial x_{i}} \right) - \frac{2}{3} \frac{\partial u_{k}}{\partial x_{k}} \delta_{ij} \right]$$

$$\frac{\partial \rho h_{s}}{\partial t} + \frac{\partial \rho u_{j} h_{s}}{\partial x_{j}} = \frac{dp}{dt} + \frac{\partial}{\partial x_{j}} \left(\rho D \frac{\partial h_{s}}{\partial x_{j}} \right) + Q$$

$$\frac{\partial \rho Y_{k}}{\partial t} + \frac{\partial \rho u_{j} Y_{k}}{\partial x_{j}} = \frac{\partial}{\partial x_{j}} \left(\rho D \frac{\partial Y_{k}}{\partial x_{j}} \right) + \omega_{k}$$



Numerical setup

Time: second-order Crank-Nicholson scheme

The convective terms: 2rd MUSCL scheme (TVD)

The viscous terms: second-order central differencing discretization



Validation – detonation cell size



Experimental cell width: 55mm Predicted value: 60mm



Westbrook's model (cell size: ~20mm)



Validation - Flame acceleration in a vented

duct

(P. H. Taylor and S. J. Bimson, "Flame propagation along a vented duct containing grids .," 22nd International Symposium on Combustion, August 1988, pp. 1355–1362, 1989.)









Exploratory study - shock and detonation propagation through a U-bend

S. M. Frolov, V. S. Aksenov, and I. O. Shamshin, "Shock wave and detonation propagation through U-bend tubes," Proceedings of the Combustion Institute, vol. 31, no. 2, pp. 2421–2428, Jan. 2007.





Numerical setup

- The geometry and set up mimics that of Frolov et al.'s experiments.
- The grid size is 0.25mm (8 grids in half reaction length) total grid number is 2.15M (10562×204)
- Six cases as listed below

Cases	Initial Pressure	Initial Tempreature
1	150 atm	2500K
2	100 atm	2500K
3	85 atm	2500K
4	75 atm	2500K
5	65 atm	2500K
6	60 atm	2500K



Results

The effects of the U-Bend:

- (1) First decelerating and then accelerating (Cases1 and 2)
- (2) First accelerating and then decelerating (Case 3)
- (3) Continuously decelerating (Case 4)
- (4) Decelerating shock wave followed separately by a flame (Cases 5 and 6)



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Case-1: First decelerating and then accelerating



Case 1: First decelerating and then accelerating



Case 2: First decelerating and then accelerating





Case 3: First accelerating and then decelerating



Case 4: Decelerating





Case 5: Decelerating shock wave followed separately by a flame



Case 6: Decelerating shock wave followed separately by a flame



Conclusions

- A new single-step reaction mechanism has been developed for propane-air mixture, covering the entire spectrum covering flame acceleration, transition to detonation and detonation.
- For the vented duct case, the predicted flame front is in good agreement with the measurements.
- For the six cases in the U-tube, the effects of the bend depend on the initial pressure. For the pressure range considered from 60 to 140 bar, four modes are predicted:
 - First decelerating and then accelerating
 - ➢ First accelerating and then decelerating
 - Continuously decelerating
 - Decelerating shock wave followed separately by a flame



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